## Intelligent Engine Systems

# Active Noise Control using Combustion Driven Actuators

Lead: Krish Ahuja

Collaborators/Advisees: Drs. Funk, Glezer, Parekh, Neumier

**Project Duration: 5 Years** 



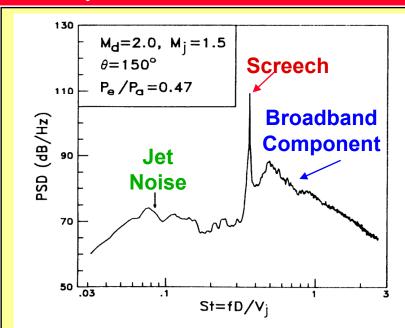




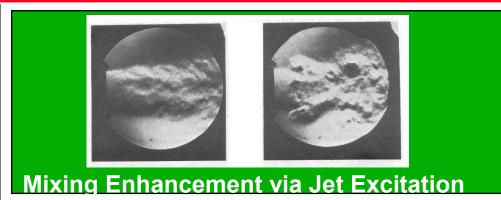


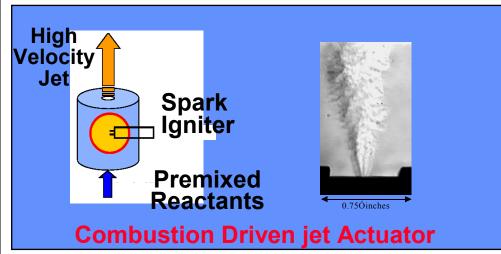
#### **Motivation and Objectives**

- Aero-engine propulsion system noise continues to be a show stopper for the aviation industry
- Innovative techniques needed to reduce noise of the current and future propulsion systems
- Techniques that have little effect on engine performance and are easily implemented are badly needed



- Jet Mixing Noise reduces by 24 dB for each halving the jet velocity
- Supersonic jet noise stands to gain most by velocity reduction













#### **Approach**

The basis of the approach lies in the fact that enhanced mixing supersonic jets reduces the extent of the supersonic flow and hence the mixing noise, the shock associated noise, and the Mach wave radiation

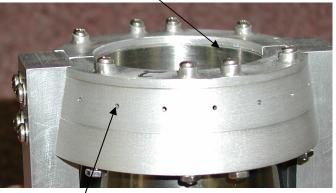
**Key: Optimized Mixing Enhancement Performance** 

The approach will build upon the results of an existing 9-month feasibility-study grant on similar work from Glenn for a round nozzle

- Step 1- Use lessons from collaborative studies to fine-tune actuator performance
- Step 2 -Test unheated nozzles
  - Open loop control
- Step 3 Implement closed loop control

Round and Rectangular Nozzles

Actuator slots



Spark holes









### Milestones and Expected Accomplishments

- Data for round and a rectangular nozzle in an anechoic chamber using an aeroacoustically clean jet flow facility
- Mixing and noise performance under each step
- Mixing characteristics and farfield acoustic signatures as a function of actuator frequency and amplitude
- Data first with open loop then with closed loop
- Correlation of the data where possible with theoretical models
- Although the focus will be on supersonic flows, the effects on subsonic jets will also be studied

Task	Yr	· 1	Yr	2	Yr	3	Yr	4	Yr	5
Adapt Nozzle Facility/Actuators										
Open Loop Test/Controller Development										
Closed Loop/Test Optimization										









#### **Chances of Success**

- We expect major challenges in producing high frequency actuation needed for model scale nozzles, especially at high jet speeds
- Excitation frequency increases with increasing jet speed and decreasing nozzle diameter
- An existing (Phase 1) NASA Glenn NRA program of feasibility study will set the course!
- Air Pulsing experiments on large-scale jets appear very promising
- Chances of success high, but after a few major hurdles and creative solutions







